

# **Data-Driven Koopman Based Fractional Order PID Control of a MEMS Gyroscope Using Bat Algorithm**

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**Abstract:** Data-driven control methods provide robust techniques for controlling the systems with a non-linear dynamic model. This paper uses the Koopman operator to linearize the non-linear dynamic model. Generating the Koopman operator is the critical part of using the Koopman theory. First, dynamic mode decomposition (DMD) is used to obtain eigenfunctions for producing the Koopman operator. Then, a fractional order PID (FOPID) controller is applied to control the linearized dynamic model. Finally, a swarm intelligence bat optimization algorithm is utilized to tune the FOPID controller's parameters. In comparison to Micro-electromechanical systems (MEMS) gyroscope controlled by a conventional PID controller, FOPID, Koopman-based FOPID controller (Koopman-FOPID), and Koopman-FOPID control optimized by bat algorithm (Koopman-BAFOPID) show better performance in comparison with three other controllers in terms of high tracking performance, low tracking error, and low control efforts.

**Keywords:** Bat algorithm, Fractional PID control, Koopman operator, Dynamic mode decomposition, MEMS gyroscope, Data-driven method.

## **1. Introduction**

The 2D MEMS gyroscope can measure angular velocities in the x and y directions. This tool has been used in the automotive industry due to its low costs and small size [1, 2]. However, controlling the MEMS gyroscope is problematic because it is constantly subjected to external disturbances.

An exciting area of study is the data-driven control system, which uses a data-driven model to control the dynamical systems [3-5]. Several techniques are used to create data-driven structures, such as deep neural networks [6] and machine learning algorithms [7]. In data-driven control systems, Koopman theory is a robust approach. By projecting the system dynamics onto the Koopman eigenspace, Goswami and Paley [8] explore the issues of bilinearization and optimum control of a control-affine non-linear system. Under certain assumptions, the suggested technique

converts the dynamics into a bilinear system using the Koopman canonical transform, especially the Koopman eigenfunctions of the drift vector field. Numerous examples of control-affine non-linear systems are used to numerically demonstrate bilinearization and the best control strategy while assuming a quadratic cost function for the states and control input. However, models created using model-based control design methodologies make it challenging to operate soft robots precisely. Koopman operator theory provides a framework for creating explicit dynamical models of soft robotics and controlling them with practiced model-based techniques [9]. The most crucial aspect of the Koopman theory is how to derive the Koopman operator, especially for complex systems with non-linear dynamic systems.

A helpful approach for estimating the modes and eigenvalues of the Koopman operator is the DMD method. Utilizing an embedding into infinite dimensional space, the Koopman operator offers a linear description of the non-linear system. The most often used finite dimensional approximations of the Koopman Operator are DMD and Extended DMD [10,11]. Koopman operator theory and the associated algorithm DMD were introduced by Ling et al. for the study and control of signalized traffic flow networks. They studied DMD's application to various issues in signalized traffic as a model-free method for describing complicated oscillatory dynamics from observed data [12]. Wilches-Bernal et al. [13] proposed a novel technique for identifying faults and other power quality issues. Their study demonstrated how one might detect events using current and voltage data to discriminate between different failures. The performance of the strategy is examined concerning the impact of the window size because the suggested method is window-based. To properly control the system, an appropriate controller can be used with the linearized DMD Koopman model, such as a linear quadratic regulator controller (LQR) [14] and a model predictive controller (MPC) [15].

PID controller is a powerful control method to control linear dynamic systems. It has been widely used in real-world scenarios due to its low cost and ease of implementation [16, 17]. It is used to control different systems, such as MEMS gyroscopes [18], vehicles [19], and quadcopters [20]. However, the main drawbacks of the PID control method are that it's not robust control against external disturbances. Fractional control is a suitable controller to improve the robustness and stability of the PID controller. FOPID controller has been used in other research. Significant uncertainty in dynamic and hydrodynamic properties and the signal transmission channel's time

delay are the key challenges with autonomous underwater vehicles (AUV) motion control. For an AUV yaw control system, Liu et al. suggested a reliable FOPID controller architecture [21]. Erol [22] proposed a method using the pitch control system of large wind turbines with a FOPID delay-dependent controller. The simulation findings demonstrate that outcomes for the delay margin are improved by using the proposed method. One of the main parts of FOPID controller design is tuning the controller's gains to achieve the best performance. The bat optimization algorithm is suitable for adjusting the proposed controller's parameters.

The bat algorithm is inspired by the behavior of microbats that use echolocation, with varying pulse rates of emission and loudness [23]. Based on the Equivalent Transfer Function model and a reduced decoupler, the [24] describes a technique for developing independent FOPID controllers for two interacting conical tank level processes. In addition, an optimization bat algorithm was used to increase the power system stability by tuning FOPID controller parameters in reference [25].

This research paper proposes a new data-driven control algorithm to control the MEMS gyroscope. The contributions of this work are as follows:

- 1- The non-linear dynamic model of the MEMS gyroscope is presented.
- 2- DMD method is used to generate eigenfunction and eigenvectors to obtain Koopman operator.
- 3- Using the Koopman theory, a FOPID controller is implemented to control the linearized dynamic model.
- 4- A bat metaheuristic optimization algorithm is used to tune the proposed control method parameters.

The rest of this paper is arranged as follows. Section two discusses the dynamic model of the MEMS gyroscope. Section three presents the Koopman theory. Section four describes the DMD method. Section five explains PID and FOPID control methods. Section six discusses the bat algorithm. Section seven provides the simulation results. Finally, section eight presents the conclusion.

## 2- Non-linear dynamic model of MEMS gyroscope

The MEMS gyroscope is a sensor for angular velocity measurement using x and y motion [26-28]. This device has been used in the automotive industry. A typical z-axis MEMS gyroscope architecture is shown in Figure 1.

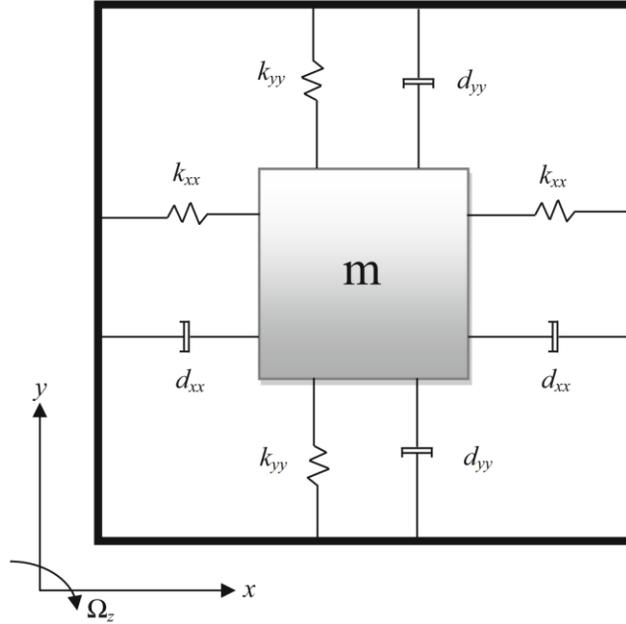


Fig. 1. MEMS gyroscope structure [21].

A typical MEMS gyroscope design includes sensor mechanisms, a proof mass suspended by springs, and an electrostatic actuation system for generating an oscillatory motion and determining the position and speed of the proof mass [29]. The proof mass is mounted on a frame that moves with a consistent linear velocity while the gyroscope rotates at a gradually varying angular velocity,  $\Omega_z$ . The centrifugal forces  $m\Omega_z^2x$  and  $m\Omega_z^2y$  are expected to be insignificant due to the modest displacements  $x$  and  $y$ . The development of the Coriolis forces,  $2m\Omega_z^*\dot{y}$  and  $2m\Omega_z^*\dot{x}$ , is parallel to the driving and rotating axes [30]. The following equations describe the dynamics of the gyroscope.

$$m\ddot{x} + d_{xx}^*\dot{x} + d_{xy}^*\dot{y} + k_{xx}^*x + k_{xy}^*y + \beta x^3 = u_x^* + 2m\Omega_z^*\dot{y} \quad (1)$$

$$m\ddot{y} + d_{xy}^*\dot{x} + d_{yy}^*\dot{y} + k_{xy}^*x + k_{yy}^*y + \beta y^3 = u_y^* - 2m\Omega_z^*\dot{x} \quad (2)$$

The origin of the coordinates in equations 1 and 2 is placed in the center of the proof mass since there is no external force applied to the system. The constants  $k_{xy}^*$  and  $d_{xy}^*$ , respectively, stand in for the asymmetric spring and damping coefficients. Despite the possibility of minor unknown deviations from their nominal values, the control forces in the x- and y-direction,  $u_x^*$  and  $u_y^*$ , are usually accepted. There are also typical descriptions of the damping rates,  $d_{xx}^*$  and  $d_{yy}^*$  and the spring constants of springs interacting in the x- and y-directions,  $k_{xx}^*$  and  $k_{yy}^*$ . Therefore, both electro-mechanical and mechanical nonlinearity, which is a positive constant, will introduce the terms  $\beta x^3$  and  $\beta y^3$ . The following vector representation might be used to express equations 1 and 2:

$$\frac{\ddot{q}^*}{q_0} + \frac{D^*}{m\omega_0} \frac{\dot{q}^*}{q_0} + \frac{K_a}{m\omega_0^2} \frac{q^*}{q_0} + \beta \frac{q^{*3}}{q_0} = \frac{u^*}{m\omega_0^2 q_0} - 2 \frac{\Omega^*}{\omega_0} \frac{\dot{q}^*}{q_0} \quad (3)$$

where

$$q^* = \begin{bmatrix} x^* \\ y^* \end{bmatrix}, \quad u = \begin{bmatrix} u_x^* \\ u_y^* \end{bmatrix}, \quad \Omega^* = \begin{bmatrix} 0 & -\Omega_z^* \\ \Omega_z^* & 0 \end{bmatrix}, \quad D^* = \begin{bmatrix} d_{xx}^* & d_{xy}^* \\ d_{xy}^* & d_{yy}^* \end{bmatrix}, \quad K_a = \begin{bmatrix} k_{xx}^* & k_{xy}^* \\ k_{xy}^* & k_{yy}^* \end{bmatrix} \text{And non-}$$

dimensional parameters are as follows:

$$q = \frac{q^*}{q_0} \quad d_{xy} = \frac{d_{xy}^*}{m\omega_0} \quad \Omega_z = \frac{\Omega_z^*}{\omega_0} \quad (4)$$

$$u_x = \frac{u_x^*}{m\omega_0^2 q_0} \quad u_y = \frac{u_y^*}{m\omega_0^2 q_0} \quad (5)$$

$$\omega_x = \sqrt{\frac{k_{xx}^*}{m\omega_0^2}} \quad \omega_y = \sqrt{\frac{k_{yy}^*}{m\omega_0^2}} \quad \omega_{xy} = \frac{k_{xy}^*}{m\omega_0^2} \quad (6)$$

where each axis' natural frequency is  $\omega_0$  and the reference length is  $q_0$ .

The following are the dynamic equations for the MEMS gyroscope.

$$\ddot{q} = -(D + 2\Omega)\dot{q} - K_b q - \beta q^3 + u + E \quad (7)$$

An external disturbance,  $E$ , might be modeled as:

$$\ddot{q} = -Y\dot{q} - Pq - \beta q^3 + u + E \quad (8)$$

where  $Y$  and  $P$  determine certain parameter variation uncertainties,  $P = K_b$  and  $Y = (D + 2\Omega)$ . Therefore, Eq. (8) might be expressed as:

$$\ddot{q} = -(Y + \Delta Y)\dot{q} - (P + \Delta P)q - \beta q^3 + u + E \quad (9)$$

where

$$q = \begin{bmatrix} x \\ y \end{bmatrix}, u = \begin{bmatrix} u_x \\ u_y \end{bmatrix}, \Omega = \begin{bmatrix} 0 & -\Omega_z \\ \Omega_z & 0 \end{bmatrix}, D = \begin{bmatrix} d_{xx} & d_{xy} \\ d_{xy} & d_{yy} \end{bmatrix}, K_b = \begin{bmatrix} \omega_x^2 & \omega_{xy} \\ \omega_{xy} & \omega_y^2 \end{bmatrix}$$

There are several ways to show Eq. (9):

$$\ddot{q} = -Y\dot{q} - Pq - \beta q^3 + u(t) + D(t) \quad (10)$$

$D(t)$  describes as:

$$D(t) = -\Delta Y\dot{q} - \Delta Pq + E \quad (11)$$

The expression for Eq. (10) in the x and y directions is

$$\begin{bmatrix} \ddot{x} \\ \ddot{y} \end{bmatrix} = - \left( \begin{bmatrix} d_{xx} & d_{xy} \\ d_{xy} & d_{yy} \end{bmatrix} + \begin{bmatrix} 0 & -2\Omega_z \\ 2\Omega_z & 0 \end{bmatrix} \right) \begin{bmatrix} \dot{x} \\ \dot{y} \end{bmatrix} - \begin{bmatrix} \omega_x^2 & \omega_{xy} \\ \omega_{xy} & \omega_y^2 \end{bmatrix} \begin{bmatrix} x \\ y \end{bmatrix} - \begin{bmatrix} \beta & 0 \\ 0 & \beta \end{bmatrix} \begin{bmatrix} x^3 \\ y^3 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} u_x \\ u_y \end{bmatrix} + \begin{bmatrix} D(t)_x \\ D(t)_y \end{bmatrix} \quad (12)$$

Eq. (12) will be transformed into first-order dynamic equations by selecting the following parameters:

$$\begin{cases} x = z_1 \\ \dot{x} = z_2 \\ y = z_3 \\ \dot{y} = z_4 \end{cases}$$

Then, there is

$$\begin{cases} \dot{z}_1 = z_2 \\ \dot{z}_2 = -\omega_x^2 z_1 - \beta z_1^3 - d_{xx} z_2 - \omega_{xy} z_3 + (2\Omega_z - d_{xy}) z_4 + u_{z_1} + D_{z_1} \\ \dot{z}_3 = z_4 \\ \dot{z}_4 = -\omega_{xy} z_1 - (d_{xy} + 2\Omega_z) z_2 - \omega_y^2 z_3 - \beta z_3^3 - d_{yy} z_4 + u_{z_3} + D_{z_3} \end{cases} \quad (13)$$

Equation (13) shows

$$\dot{z} = A(z) + Bu \quad (14)$$

Eq. (14) can be written in its classical form as follows:

$$\frac{d}{dt}z(t) = f(z) \quad (15)$$

### 3- Koopman theory

According to the Koopman operator theory, the crucial step to correctly a non-linear dynamical system is to transform the non-linear system's original form into an infinite dimensional state space so that the resulting system is linear [15].

The dynamic in discrete time defines as [31]:

$$z_{k+1} = F(z_k) \quad (16)$$

where F is characterized by

$$F(z(t_0)) = z(t_0) + \int_{t_0}^{t_0+t} f(z(\tau))d\tau \quad (17)$$

When a finite-dimensional non-linear system's dynamics are transferred to an infinite-dimensional function space using the Koopman operator theoretic method, the original system's dynamics become linear.  $g$  is a real-valued scalar measurement function and an observable part of an infinite-dimensional Hilbert space. The Koopman operator generates based on this observable as

$$Kg = g \circ F \quad (18)$$

Smooth dynamics can be implemented using a continuous system.

$$\frac{d}{dt}g(z) = Kg(z) = \nabla g(z) \cdot f(z) \quad (19)$$

where the Koopman operator is K. The infinite dimensions of the Koopman operator are significant but problematic for operation and representation. Instead of describing the development of all measurement functions in a Hilbert space, applied Koopman analysis approximates the evolution on a subspace covered by a small number of measurement functions. One can get a representation of the Koopman operator in a finite-dimensional matrix by limiting the operator to an invariant subspace. A Koopman invariant subspace is covered by any combination of the eigenfunctions of

the Koopman operator [31]. When eigenvalue  $\lambda$  is satisfied by eigenfunction  $\varphi(z)$  of the Koopman model.

$$\lambda\varphi(z) = \varphi(F(z)) \quad (20)$$

In continuous time, a Koopman eigenfunction  $\varphi(z)$  is satisfied.

$$\frac{d}{dt}\varphi(z) = \lambda\varphi(z) \quad (21)$$

A finite-dimensional approximation is required from the application side to approximate the Koopman operator. DMD method is one of the approaches that can estimate the Koopman operator [31].

#### 4- DMD method

A numerical DMD is utilized to approximate the Koopman operator.

$$Z' \approx AZ \quad (22)$$

where  $Z'$  is time-shifted of matrix  $Z$  as:

$$Z = [z_1 \quad z_2 \quad \dots \dots]$$

The  $A$  can be found according to Eq. (22) as:

$$A = Z'Z^+ \quad (23)$$

where  $+$  represents the Moore-Penrose pseudoinverse. We may use Singular Value Decomposition (SVD) on the snapshots to determine the dominating properties of the pseudoinverse of  $Z$  because a typical calculation involving  $A$  would need a significant amount of computation due to its size [32].

$$Z \approx U\Sigma V^* \quad (24)$$

where  $U \in R^{n \times r}$ ,  $\Sigma \in R^{r \times r}$ ,  $V \in R^{n \times r}$ , and  $*$  demonstrate the conjugate transpose. SVD's reduced rank for approximating  $Z$  is  $r$ . The eigenvectors can be defined as:

$$\Phi = Z'V\Sigma^{-1}W \quad (25)$$

where  $W$  is the eigenvectors of full-rank system dynamic systems.

$$\Phi = Z'V\Sigma^{-1}W \quad (26)$$

Let  $\lambda$  be eigenfunction, and then we will have:

$$KW = \lambda W \quad (27)$$

where  $K$  is the Koopman operator.

The linearized dynamic model can be demonstrated as:

$$\frac{d}{dt}y = Ky + Bu \quad (28)$$

## 5- FOPID control

PID controller is a suitable control method used in many industrial applications [33-35]. It constantly evaluates the error using its parameters  $K_p$ ,  $K_i$ , and  $K_d$  and delivers the correct value.

The PID controller can be defined as:

$$u_{PID} = K_p e(t) + K_i \int_0^t e(\tau) d\tau + K_d \frac{de(t)}{dt} \quad (29)$$

where  $e(t) = y - y_d$ , which  $y_d$  is the desired trajectory.

The main problem of the PID controller is that it's not robust against external disturbances. Also, the stability of the PID controller is another issue that should be considered during the controller design.

A fractional control method can be used to improve the controller's performance. In addition, it can improve the stability and robustness of a typical PID controller. The FOPID controller can be defined as:

$$u_{FOPID} = K_p e(t) + K_i D^{-\mu} e(t) + K_d D^{\mu} e(t) \quad (30)$$

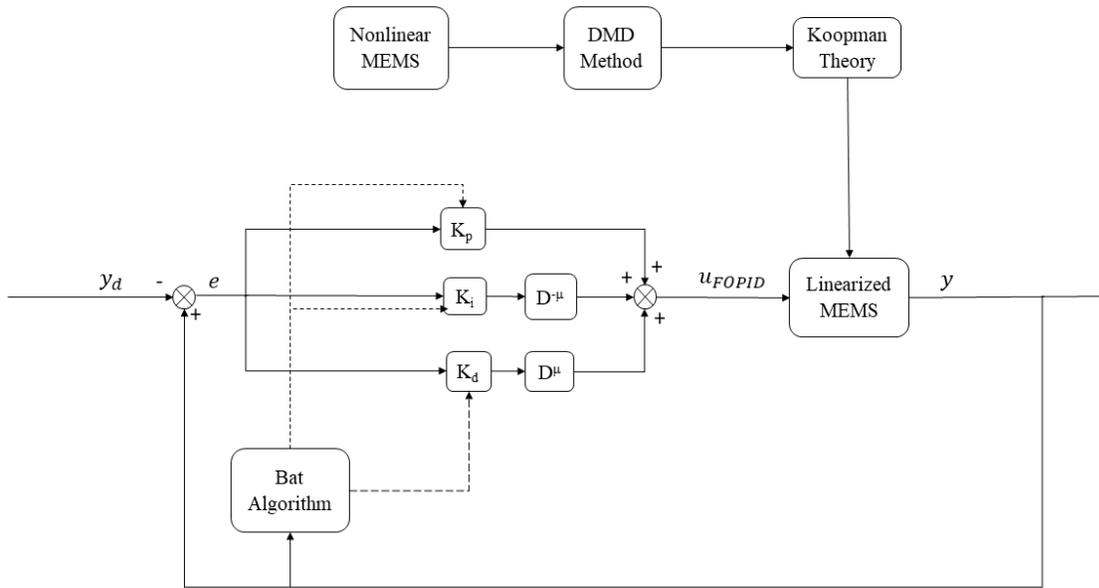
where  $D$  is a fractional operator defined as  $D = \frac{d}{dt}$  and  $\mu$  is fractional order. The fractional type that we use in this research is Grunwald-Letnikov [36]. The Grunwald-Letnikov fractional derivative of the function  $e(t)$  with respect to  $t$  is given

$$D_t^\mu e(t) = \lim_{h \rightarrow 0} h^{-\mu} \sum_{k=0}^n (-1)^k \binom{\mu}{k} f(e(t) - kh) \quad (31)$$

where

$$\binom{\mu}{k} = \frac{\mu(\mu-1)(\mu-2)\dots(\mu-k+1)}{k!} = \frac{\Gamma(\mu+1)}{k! \Gamma(\mu-k+1)}$$

A detailed explanation is given in [36]. The control structure shows in Fig. 2.



**Fig. 2.** The proposed control structure.

One of the main parts of FOPID controller design is how to tune the controller's parameters. The metaheuristic algorithms are rich sources to adjust the FOPID controller parameters.

### 6- Bat algorithm to tune the proposed controller parameters

The optimization technique known as the bio-inspired bat algorithm was influenced by how bats use echolocation to find food. It was introduced in [37, 38] and used to resolve several optimization issues. The echolocation strategy of bats is used in the algorithm. These bats create an extremely loud sound pulse; then, they listen for the echo returned from the nearby objects. Depending on

the species, their signal bandwidth ranges change through harmonics. The  $i^{th}$  bat moves randomly at location  $x_i$  with velocity  $v_i$  and a set frequency  $f_{min}$ . To discover food, the bat changes its wavelength and volume. To improve the echolocation capabilities, an objective function needs to be optimized. Developing an optimization algorithm from how a bat searches for the best answer is possible. The algorithms for bat-inspired echolocation can be created by enhancing certain of the microbats' echolocation characteristics. The bat echolocation features to address an optimization issue brought about by the following hypotheses [39].

- 1- An echolocation is a tool used by all bats to detect distance.
- 2- To locate prey, bats fly at random speeds of  $v_i$  at positions  $x_i$  with a fixed frequency and wavelength of  $f_{min}$  and a variable wavelength and frequency of  $A_0$ .
- 3- Depending on how close the prey is, they can control their wavelength/frequency and pulse emission rate,  $r_i \in [0-1]$ .
- 4- Their loudness decreases from high  $A_0$  to low  $A_{min}$  levels as they get closer to the prey.

In actual implementations, frequency occurs between  $[f_{min}, f_{max}]$  and is chosen to be similar to the size of the domain of interest. For a virtual bat to solve an optimization issue, rules must be developed to specify their locations and velocities in the d-dimensional search space. The following definitions apply to the new location  $x_i^+$  and velocity  $v_i^+$  at time step  $t$  [40].

$$f_i = f_{min} + (f_{max} - f_{min})\xi \quad (32)$$

$$v_i^t = v_i^{t-1} + (x_i^{t-1} - x^*)f_i \quad (33)$$

$$x_i^t = x_i^{t-1} + v_i^t \quad (34)$$

The current best solution across all  $N$  bats is represented by  $x^*$ , where  $\xi \in [0-1]$  is the random vector generated randomly from a uniform distribution. When a new solution is needed for local search, it is determined using the most recent bat loudness  $A_i$  and the most variance that can be tolerated  $\max(\text{var})$  at a time stop, as shown below.

$$x_{new} = x_{old} + \epsilon A_i \max(\text{var}) \quad (35)$$

The volume drops, and the pulse emission rate rises as a bat locates its prey. The bat is heading towards the target, as shown by

$$A_i^{t+1} = \alpha A_i^t, \quad r_i^{t+1} = r_i^0 [1 - e^{-\gamma t}] \quad (36)$$

Where  $\alpha$  and  $\gamma$  are constant. Initial boundness is  $A_i \in [0.1-0.9]$ , initial emission rate is  $r_0 \in [0-1]$ , and  $\alpha = \gamma = 0.9$ . The bat algorithm is used for tuning the  $[K_p, K_i, K_d]$  parameters of the proposed controller for a MEMS gyroscope. This problem's objective function is described as follows [16]:

$$J = \int_0^\infty (w_1 |e(t)| + w_2 u^2(t)) dt + w_3 t_u \quad (37)$$

## 7- Simulation results

A MEMS gyroscope is controlled using the proposed Koopman-BAFOPID controller. Additionally, several comparative methods are used to show how effective the proposed Bat algorithm is in adjusting the Koopman-FOPID parameters. The non-linear dynamic equations of a MEMS gyroscope are generated in this research. All simulation steps are simulated using Matlab software. Fig. 3 shows the flow chart of the bat algorithm steps in tuning the Koopman-FOPID controller. The parameters of the proposed controller in this study are as follows: total population = 5; iteration = 20; loudness = 0.5; wavelength = 0.5; frequency  $f_{min} = 10, f_{max} = 20$ .

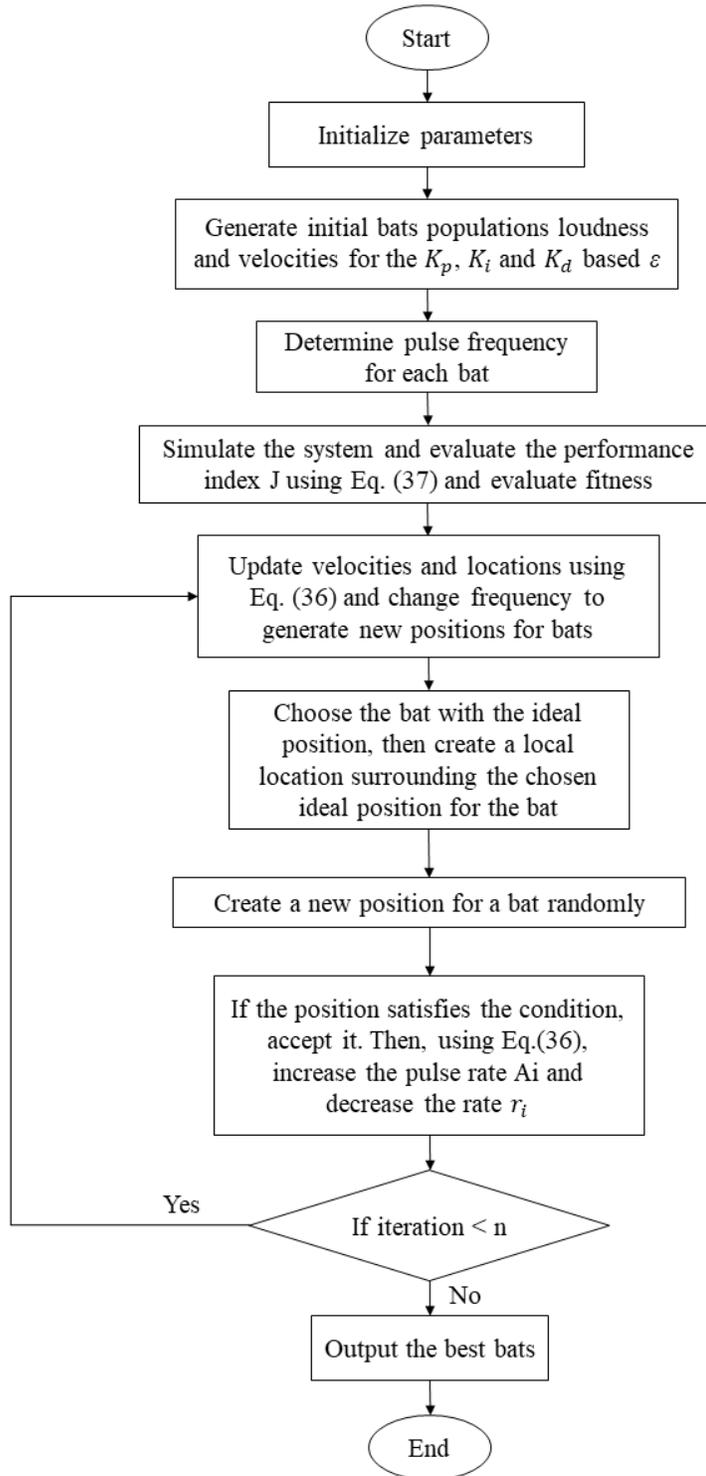
The objective function variables are set to  $w_1=0.99, w_2=0.01, w_3=2$ . The tuned parameters of bat algorithm are  $K_p = \text{diag}\{17.9901\}, K_i = \{22.3411\},$  and  $K_d = \{27.2585\}$ .

Fig. 4 shows the trajectory tracking of x and y direction under PID, FOPID, Koopman-FOPID, and Koopman-BAFOPID controllers. It demonstrates that the proposed Koopman-BAFOPID controller has a high tracking performance compared to the three other controllers. Fig. 5 shows the position tracking error of x and y directions under PID, FOPID, Koopman-FOPID, and Koopman-BAFOPID controllers. It illustrates that the proposed Koopman-BAFOPID controller has low tracking error in comparison with the PID, FOPID, Koopman-FOPID, and Koopman-BAFOPID controllers. Fig. 6 show the velocity of x and y direction under PID, FOPID, Koopman-FOPID, and Koopman-BAFOPID controllers. First, a conventional PID controller is applied to a non-linear MEMS gyroscope to control the x and y direction. The main problem with that controller is that it's not stable. Then, a FOPID controller is used to remove conventional PID controllers' stability problem. Fig. 7 shows the control inputs under PID, FOPID, Koopman-FOPID, and Koopman-BAFOPID controllers. It demonstrates that the PID controller is unstable by increasing the control efforts when time increases, but the FOPID controller fixed this problem.

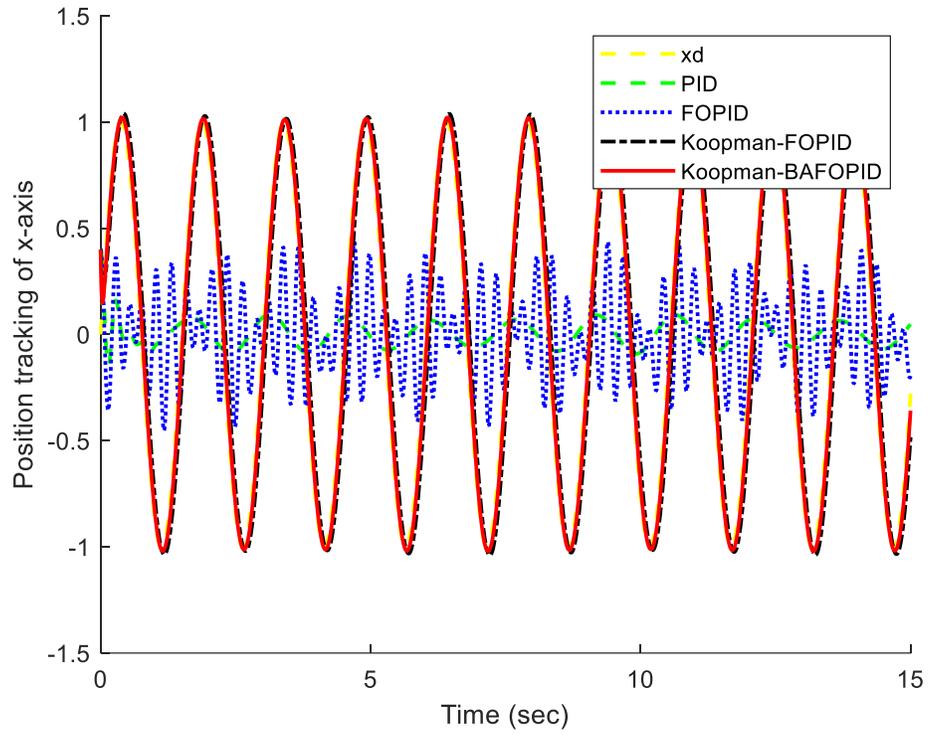
Therefore, using the FOPID controller provides better stability, along with using Koopman theory on a non-linear MEMS gyroscope.

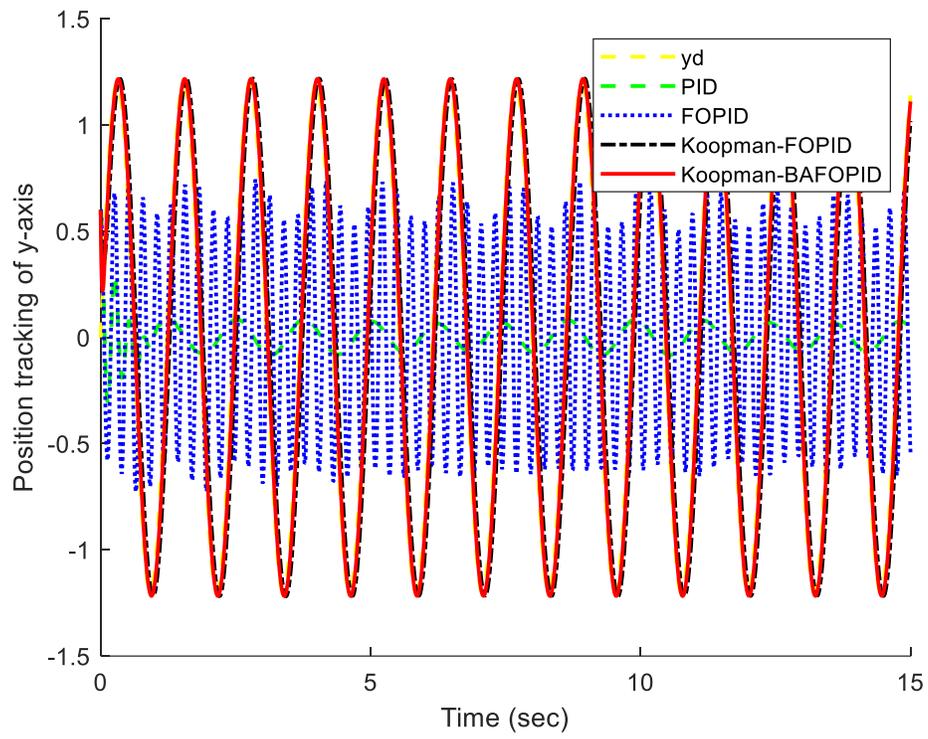
## **8. Conclusion**

This paper proposed a new Koopman-BAFOPID control of a non-linear MEMS gyroscope. The PID controller stability improved by presenting the FOPID controller. The Koopman theory is used to derive a linear dynamic model of the MEMS gyroscope. The DMD method was used to estimate the Koopman operators numerically. Then, the selected FOPID controller was then applied to a linearized MEMS gyroscope dynamic model to suitably controlling the x and y direction. A bat algorithm was implemented on the Koopman-FOPID controller to tune the proposed controller parameters. The simulation results verified that the proposed Koopman-BAFOPID controller has better performance than PID, FOPID, and Koopman-FOPID controllers in terms of high tracking performance, low tracking error, low control efforts, and increased stability.

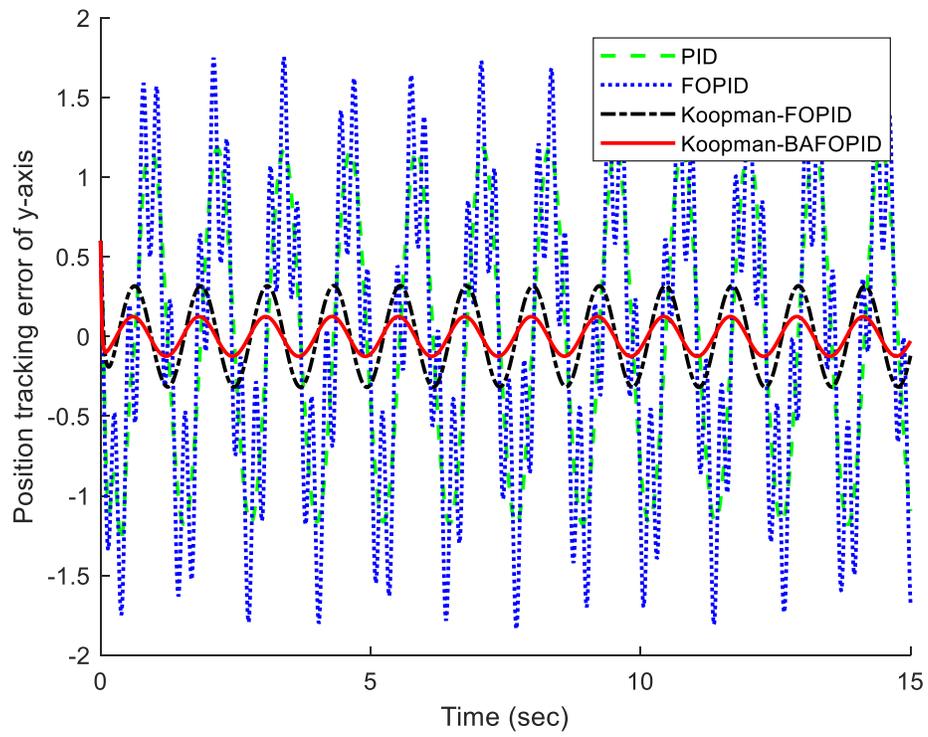
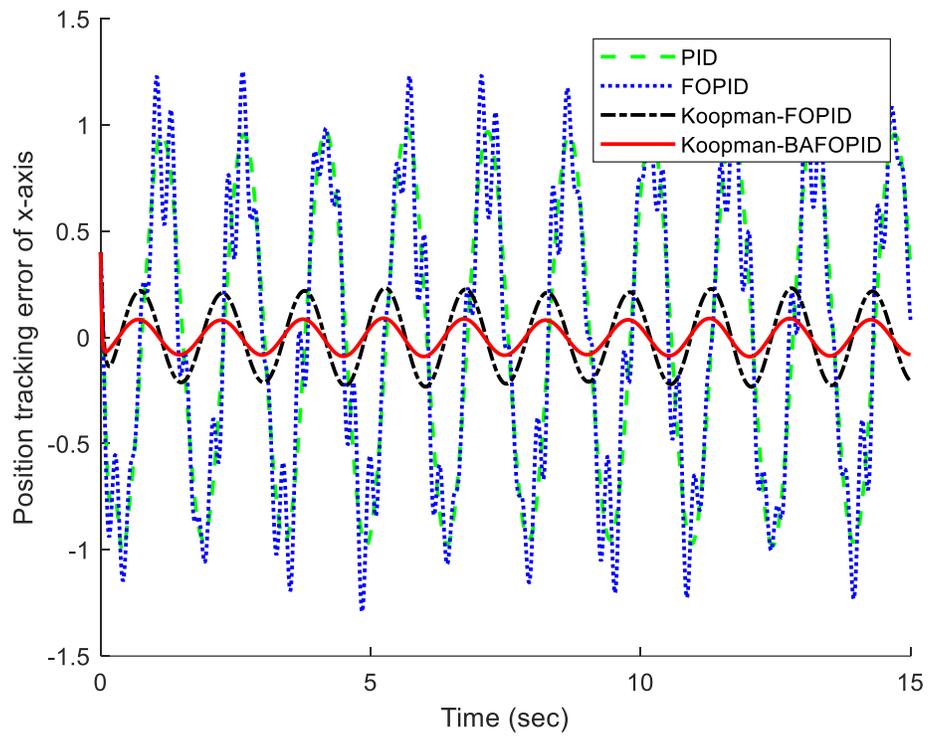


**Fig. 3.** Flow chart for a bat algorithm for tuning Koopman-FOPID parameters.

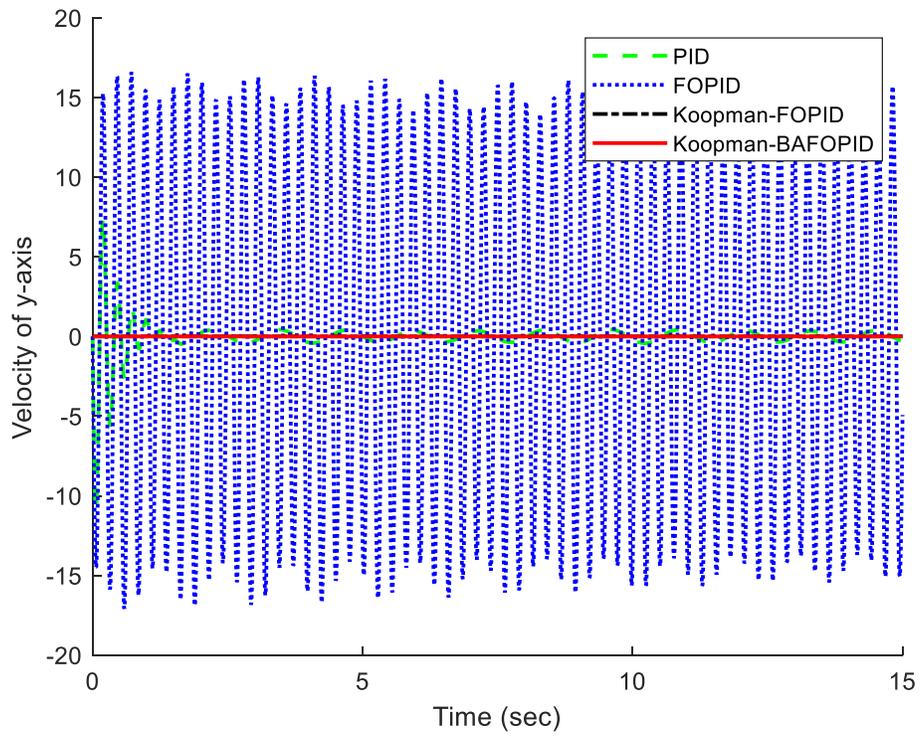
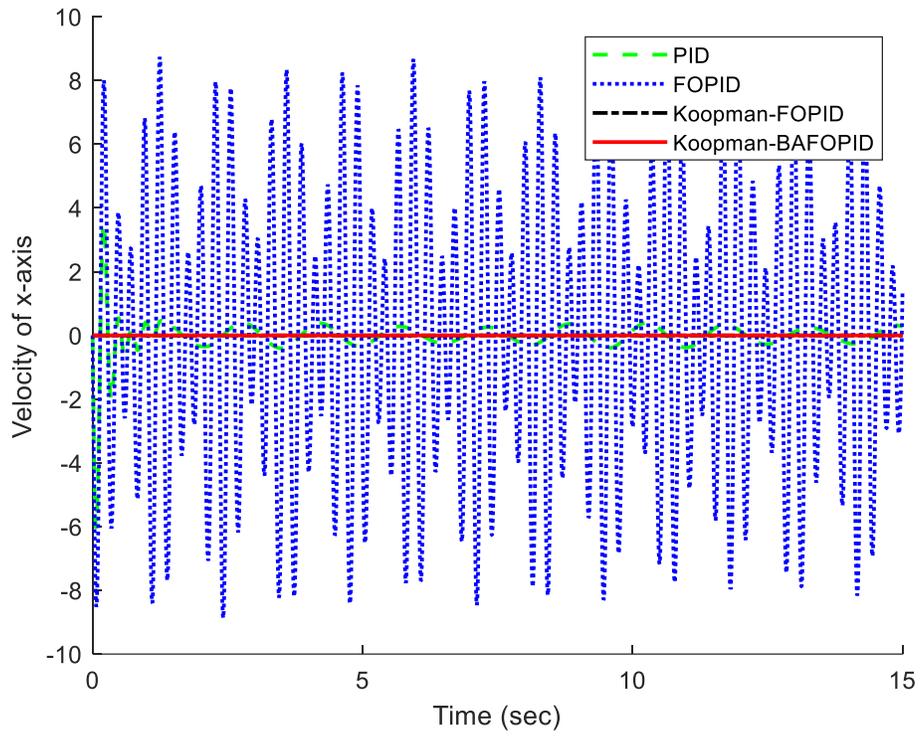




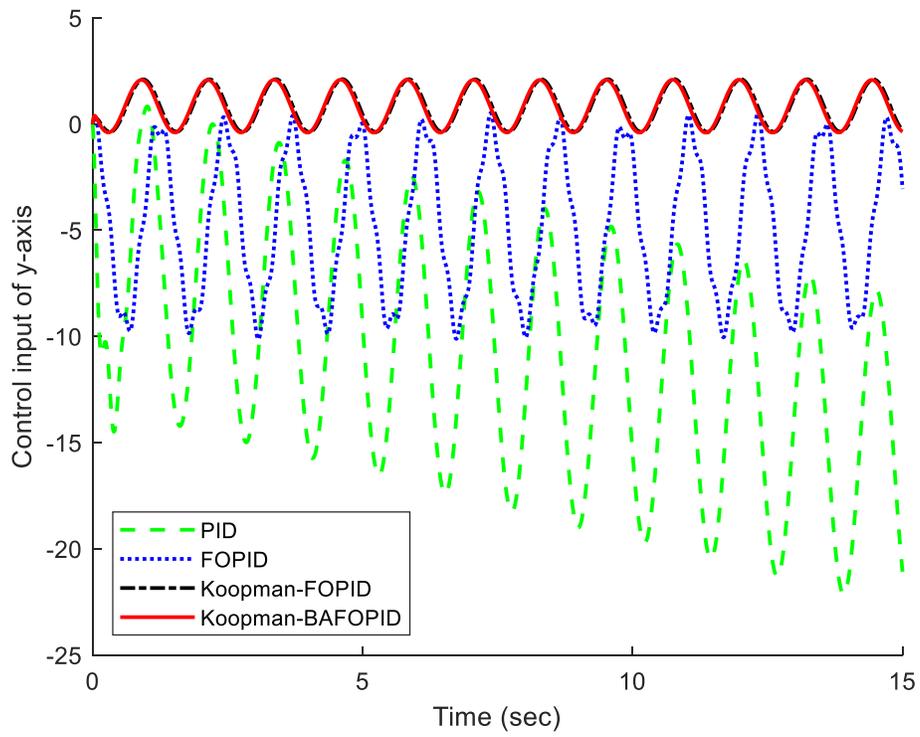
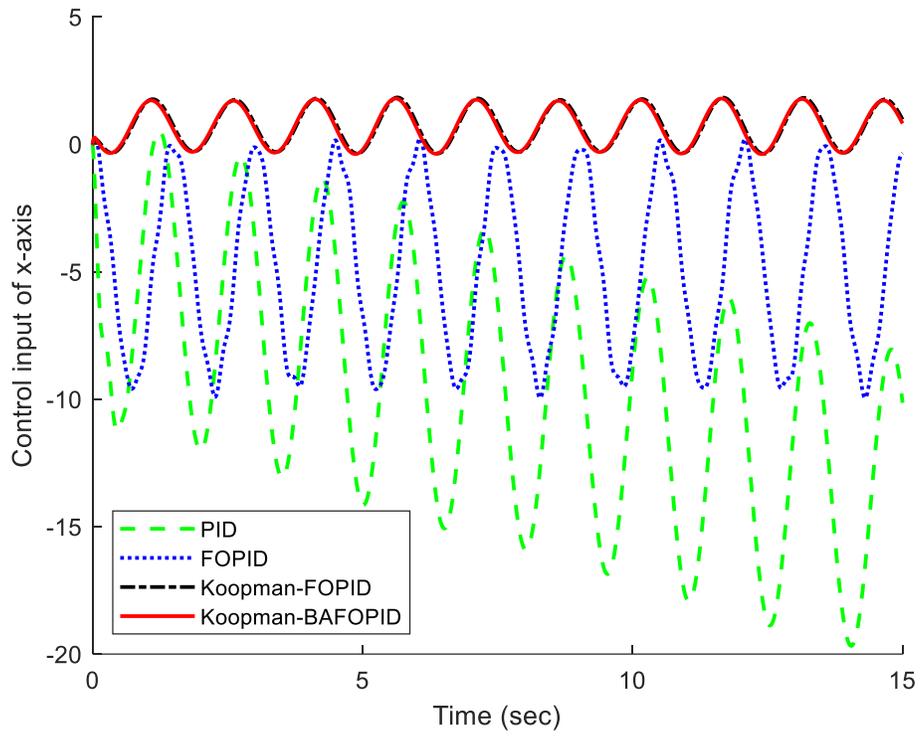
**Fig. 4.** Position tracking of x and y under the proposed controllers.



**Fig. 5.** Position tracking error of x and y directions under the proposed controllers.



**Fig. 6.** Velocity of x and y directions under the proposed controllers.



**Fig. 7.** Velocity of x and y directions under the proposed controllers.

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